

# ADJOINTS AND ANISOTROPIC MESH ADAPTATION FOR COMPRESSIBLE GAS DYNAMICS

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with the participation of: A. Belme, A. Dervieux, A. Loseille

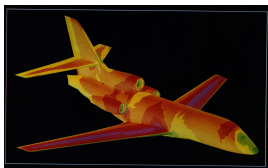
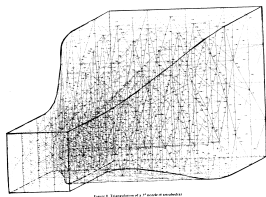
1. INRIA - Gamma3 Project - Rocquencourt, France
2. University of Paris VI, Laboratoire J.-L. Lions, France

San Diego, 22 Juin 2013

- ① History, Context and Motivations
- ② Metric-Based Mesh Adaptation for Steady Flows
- ③ Metric-Based Mesh Adaptation for Unsteady Flows

# Early CFD (as recollected by O. Pironneau)

- 1972 Unstructured meshes introduced at Dassault by Pierre Perrier
- 1973 First FEM code runs in 3D (J. Periaux et al)
- 1976 Transonic flow as an abstract least square and use conjugate gradient
- 1977 Meeting with Antony Jameson at the von Karman Institute
- 1978 Full potential flow computation around a flacon jet
- 1983 Van Leer & Phil Roe visits INRIA-Sophia and work with Alain Dervieux
- 1987 Hermes Program



M. O. Bristeau, O. Pironneau, R. Glowinski, J. Periaux, and P. Perrier, On the numerical solution of nonlinear problems in fluid dynamics by least squares and FEM. *Comput. Methods Appl. Mech. Engrg.*, Vol. 17/18(part 3), pp. 619-657, 1979

## Numerical Simulation Pipeline

CAD → **MESH** → SOLVER → VISU / ANALYSIS

① *no mesh = no simulation*

② *a "bad" mesh implies a wrong or inaccurate solution*

## Automated Unstructured Tetrahedra Mesh Generation Methods:

- Octree-like [Yerry and Shephard, IJNME 1984], ...
- Advancing front [Lohner and Parikh, IJNMF 1988], [Peraire et al., IJNME 1988], [Jin and Tanner, IJNME 1991], ...
- Delaunay [Hermeline, RAIRO AN 1982], [Baker, AIAA 1987], [George, Hecht and Saltel, ICSE 1990], [Weatherhill, CMA 1992], ...
- Minimal volume [Coupez, REEF 2000], ...
- Coupled Delaunay-frontal [Marcum and Weatherhill, AIAA 1995], ...

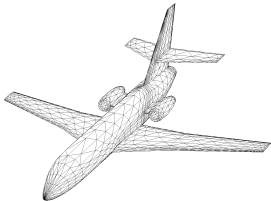
At the end of 90's

**3D** powerful and mature mesh generation methods become available

## Numerical Simulation Pipeline

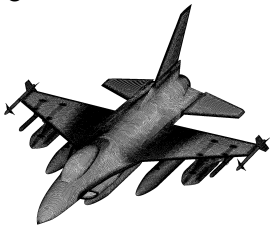
CAD  $\rightarrow$  **MESH**  $\rightarrow$  SOLVER  $\rightarrow$  VISU / ANALYSIS

- 1 *no mesh = no simulation*
  - 2 *a "bad" mesh implies a wrong or inaccurate solution*
- Address ever increasing **geometrical** complexity  
 $\Rightarrow$  take into account geometric features inside the mesh



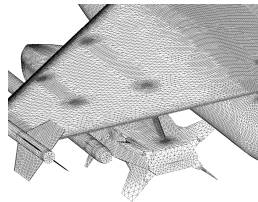
1990

$h_{surf} = 10 \text{ cm}$



2000

$h_{surf} = 1 \text{ mm}$  and  $h_{BL} = 1 \mu\text{m}$

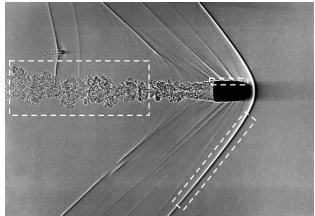


• Address ever increasing **physical** complexity

## Numerical Simulation Pipeline

CAD  $\rightarrow$  **MESH**  $\rightarrow$  SOLVER  $\rightarrow$  VISU / ANALYSIS

- 1 *no mesh = no simulation*
  - 2 *a "bad" mesh implies a wrong or inaccurate solution*
- Address ever increasing **geometrical** complexity
  - Address ever increasing **physical** complexity  
 $\implies$  take into account flow characteristics inside the mesh



## Numerical Simulation Pipeline

CAD  $\rightarrow$  **MESH**  $\rightarrow$  SOLVER  $\rightarrow$  VISU / ANALYSIS

- 1 *no mesh = no simulation*
- 2 *a "bad" mesh implies a wrong or inaccurate solution*

- Address ever increasing **geometrical** complexity
- Address ever increasing **physical** complexity
- Address **convergence studies** in 3D

$$h \rightsquigarrow N \quad \text{and} \quad dt \sim h \quad \implies \quad \text{CPU} \times \mathbf{1}$$

$$\frac{h}{2} \rightsquigarrow 8N \quad \text{and} \quad dt \sim h \rightsquigarrow \frac{dt}{2} \quad \implies \quad \text{CPU} \times \mathbf{16}$$

$$\frac{h}{4} \rightsquigarrow 64N \quad \text{and} \quad dt \sim h \rightsquigarrow \frac{dt}{4} \quad \implies \quad \text{CPU} \times \mathbf{256}$$

## Numerical Simulation Pipeline

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  - Address ever increasing **physical** complexity
  - Address **convergence studies** in 3D

Require **tailored** meshes to address and certify numerical results



**Modify** discretization of  $\Omega$  to **control** solution accuracy

## Numerical Simulation Pipeline

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## Structured boundary layer mesh generation:

- Open advancing-layer method  
[Lohner and Parikh, IJNMF1988], [Kallinderis and Ward, AIAA 1993], [Pirzadeh, AIAA 1994], [Marcum, AIAA 1995], [Sharov and Nakahashi, AIAAJ 1998], [Garimella and Shephard, IJNMF 2000], ...
- Closed advancing-layer method by pushing  
[Hassan et al, IJNME 1996], [Ito and Nakahashin, IMR 2002], [Bottasso and Detomi, 2002], ...
- Closed advancing layer method with iterative point insertion  
[Marcum, AIAA 1995], [Loseille and Lohner, AIAA 2011], ...



## Numerical Simulation Pipeline

CAD → **MESH** → SOLVER → VISU / ANALYSIS

- 1 *no mesh = no simulation*
- 2 *a "bad" mesh implies a wrong or inaccurate solution*

## Anisotropic Mesh Adaptation:

- Error measures with **directions** in 2D. Use local mapping  
[Peraire et al., JCP 1987], [Lohner, CMAME 1989], [Selmin and Formaggia, IJNME 1992], ...

In 1994, **O. Zienkiewicz** and **J. Wu**. gave a status.

Even if they had great success with such approach, they said:

*"Unfortunately the amount of elongation which can be used in a typical mesh generation by such mapping is **small**..."*

## Numerical Simulation Pipeline

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## Anisotropic Mesh Adaptation:

- Error measures with **directions** in 2D. Use local mapping [Peraire et al., JCP 1987], [Lohner, CMAME 1989], [Selmin and Formaggia, IJNME 1992], ...  
 $\implies$  But, mesh anisotropy was **small**
- Stretched elements with a **Delaunay** approach in 2D [Mavriplis, JCP 1990]
- Introduce the use of **metrics** in a 2D Delaunay mesh generator [George, Hecht and Vallet, AES 1991]

What's a metric and metric-based mesh generation ?

# Generation of Adapted Meshes

## ① How to communicate with an automatic mesh generator ?

**Main idea:** change mesh generator **distance and volume computation**

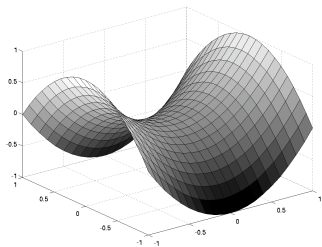
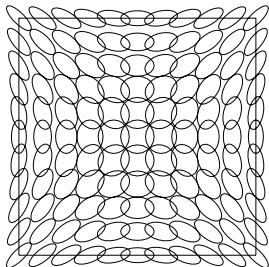
[George, Hecht and Vallet., Adv. Eng. Software 1991]

**Fundamental concept:** The notion of **metric** and Riemannian metric space

Computing geometric quantities in Riemannian metric space  $\mathbf{M} = (\mathcal{M}(\mathbf{x}))_{\mathbf{x} \in \Omega}$



Computing geometric quantities on  $\mathcal{S}$



# Generation of Adapted Meshes

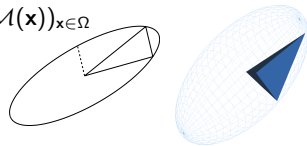
## ① How to communicate with an automatic mesh generator ?

**Main idea:** change mesh generator **distance and volume computation**

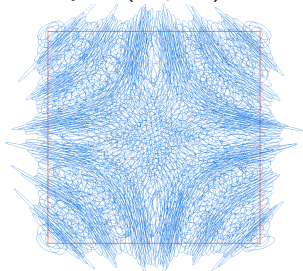
[George, Hecht and Vallet., Adv. Eng. Software 1991]

**Fundamental concept:** Generate a **unit mesh** w.r.t  $(\mathcal{M}(\mathbf{x}))_{\mathbf{x} \in \Omega}$

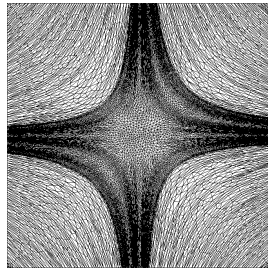
$$\forall \mathbf{e}, \ell_{\mathcal{M}}(\mathbf{e}) \approx 1 \text{ and } \forall K, |K|_{\mathcal{M}} \approx \begin{cases} \sqrt{3}/4 & \text{in 2D} \\ \sqrt{2}/12 & \text{in 3D} \end{cases}$$



**Inputs**  $(\mathcal{H}_0, \mathcal{M}_i)_{i \in \mathcal{H}}$



**Output**  $\mathcal{H}$



# Generation of Adapted Meshes

- 1 How to communicate with an automatic mesh generator ?
- 2 How to measure or quantify mesh size and anisotropy ?

Use appropriate error estimates

## Anisotropic Mesh Adaptation State-of-the-art:

The **fruitful idea of metric** was widely exploited in 2D:

[Fortin et al., ECCOMAS 1996], [Castro-Diaz et al, IJNMF 1997], [Dompierre et al., AIAA 1997], [Buscaglia and Dari, IJNME 1997], [Formaggia and Perotto, NM 2001], [Picasso, SIAMJSC 2003], ...

At the end of 90's

2D anisotropic mesh adaptation proves to be efficient in CFD

# Generation of Adapted Meshes

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## Anisotropic Mesh Adaptation State-of-the-art:

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In 1997, **T. Baker** wrote:

*"Mesh generation in three dimensions is difficult enough task in the absence of mesh adaptation and it is only recently that satisfactory three-dimensionnal mesh generators have become available. [...] . Mesh alteration in three dimensions is therefore a rather perilous procedure that should be undertaken with care"*

## Interpolation error based mesh adaptation

Deriving the **best mesh** to observe a given solution field  $W$

- Generic: does not depend on the PDE and numerical scheme
- Anisotropy is easily deduced

[Tam et al., CMAME 2000], [Pain et al., CMAME 2001], [Formaggia and Perotto, NM 2001], [Picasso, SIAMJSC 2003], [Formaggia et al., ANM 2004], [Bottasso, IJNME 2004], [Li et al., CMAME 2005], [Frey and Alauzet, CMAME 2005], [Gruau and Coupez, CMAME 2005], [Huang, JCP 2005], [Compere et al., 2007], ...

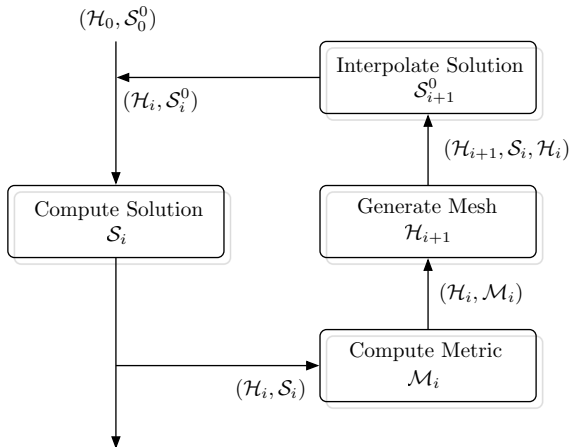
## Issues with feature-based anisotropic mesh adaptation for CFD

- **Robustness** of 3D anisotropic mesher
- **Loss of anisotropy** for non smooth solutions:  $h_{min}$  size in singularities
- Cannot capture **small scales** of the flow
- **Order one** convergence for non-smooth solutions with **2nd-order** scheme

# Anisotropic Mesh Adaptation for Steady Problems

Mesh adaptation is a **non-linear problem**

⇒ an **iterative process** is required to converge the couple mesh-solution





## Issue

no mesh  $\implies$  no computation



## Remedy

Local adaptive remesher  $\implies$  always valid

- EPIC [Michal and Krakos, AIAA 2012]
- Feflo.a [Loseille and Lohner, AIAA 2010]
- Forge3d [Coupez, REEF 2000]
- Fun3d [Jones et al., AIAA 2006]
- MAdLib [Compere et al., IJNME 2010]
- MeshAdap [Li et al., IJNME 2005]
- Mmg3d [Dobrzynski and Frey, IMR 2008]
- Mom3d [Tam et al., CMAME 2000]
- Tango [Bottasso, IJNME 2004]
- Libadaptivity [Pain et al., CMAME 2001]

# Continuous Mesh Theory

We proposed a **continuous mesh framework** to theorize mesh adaptation

[Alauzet et al., IMR 2006], [Alauzet, IJNMF 2008], [Loseille and Alauzet, SINUM 2010]

Discrete	Continuous
Element $K$	Metric tensor $\mathcal{M}$
Volume $ K $	Volume $\alpha (\det \mathcal{M})^{-\frac{1}{2}}$
Mesh $\mathcal{H}$ of $\Omega_h$	Riemannian metric space $\mathbf{M} = (\mathcal{M}(\mathbf{x}))_{\mathbf{x} \in \Omega}$
Number of vertices $N_v$	Complexity $\mathcal{C}(\mathbf{M}) = \int_{\Omega} \sqrt{\det(\mathcal{M}(\mathbf{x}))} dx$
Linear interpolate $\Pi_h u$	Continuous linear interpolate $\pi_{\mathcal{M}} u$

## Local interpolation error duality

For all  $K$  unit for  $\mathcal{M}$  and for all  $u$  quadratic positive form ( $u(\mathbf{x}) = \frac{1}{2} {}^t \mathbf{x} H_u \mathbf{x}$ ):

$$\|u - \Pi_h u\|_{L^1(K)} = \frac{\sqrt{2}}{240} \underbrace{\det(\mathcal{M}^{-\frac{1}{2}})}_{\text{mapping}} \underbrace{\text{trace}(\mathcal{M}^{-\frac{1}{2}} H_u \mathcal{M}^{-\frac{1}{2}})}_{\text{anisotropic term}} = \|u - \pi_{\mathcal{M}} u\|_{L^1(K)}$$

Working in this framework enables us to use powerful mathematical tool

## Application: Minimizing the Interpolation Error in $L^p$ -norm

- An ill-posed discrete problem

Find  $\mathcal{H}_{opt}$  having  $N$  vertices such that

$$\mathcal{H}_{opt}(u) = \text{Arg min}_{\mathcal{H}} \|u - \Pi_h u\|_{\mathcal{H}, L^p(\Omega_h)}$$

- A well-posed continuous problem

Find  $\mathbf{M}_{L^p} = (\mathcal{M}_{L^p}(\mathbf{x}))_{\mathbf{x} \in \Omega}$  of complexity  $N$  such that

$$\begin{aligned} E_{L^p}(\mathbf{M}_{L^p}) &= \min_{\mathbf{M}} E_{L^p}(\mathbf{M}) = \min_{\mathbf{M}} \|u - \pi_{\mathcal{M}} u\|_{L^p(\Omega)} \\ &= \min_{\mathbf{M}} \left( \int_{\Omega} |u(\mathbf{x}) - \pi_{\mathcal{M}} u(\mathbf{x})|^p \, d\mathbf{x} \right)^{\frac{1}{p}} \end{aligned}$$

$\implies$  Solved by a calculus of variations

Optimal metric [Alauzet et al., IMR 2006], [Loseille and Alauzet, SINUM 2010]

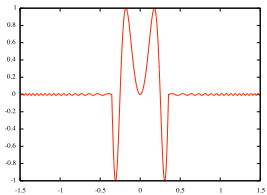
$$\mathcal{M}_{L^p}(\mathbf{x}) = N^{\frac{2}{3}} \underbrace{\left( \int_{\Omega} (\det |H_u|)^{\frac{p}{2p+3}} \right)^{-\frac{2}{3}}}_{\text{1}} \underbrace{(\det |H_u(\mathbf{x})|)^{\frac{-1}{2p+3}}}_{\text{2}} \underbrace{|H_u(\mathbf{x})|}_{\text{3}}$$

- $\mathbf{M}_{L^p}$  is unique
- $\mathbf{M}_{L^p}$  has for optimal directions and ratios the Hessian ones
- $\mathbf{M}_{L^p}(u)$  provides an optimal explicit bound of the interpolation error in  $L^p$  norm:

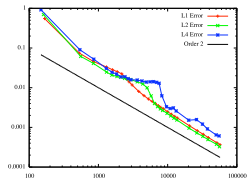
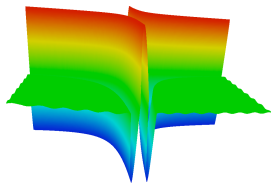
$$\|u - \pi_{\mathcal{M}_{L^p}} u\|_{L^p(\Omega)} = 3 N^{-\frac{2}{3}} \left( \int_{\Omega} (\det |H_u|)^{\frac{p}{2p+3}} \right)^{\frac{2p+3}{3p}}$$

- Global second order of convergence for a sequence of embedded continuous meshes  $(\mathbf{M}_{L^p}^N(u))_{N=1 \dots \infty}$

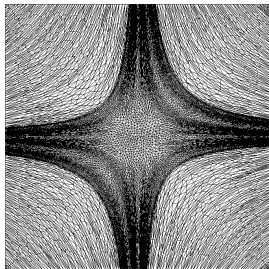
# Continuous Mesh Theory



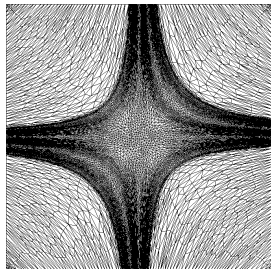
Multi-scale function



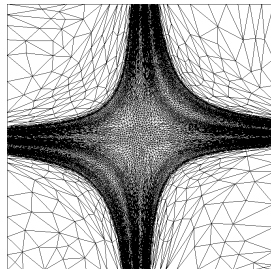
Asymptotic 2nd order cv



$L^1$ -adaptation



$L^2$ -adaptation



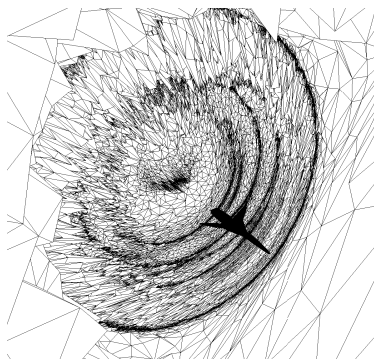
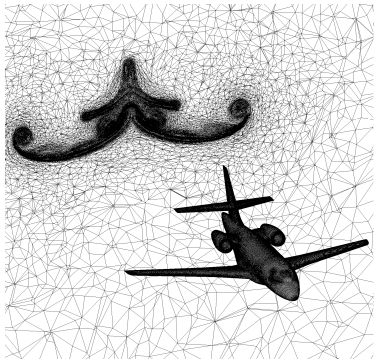
$L^4$ -adaptation

# Feature-based Anisotropic Mesh Adaptation

We propose a **multiscale anisotropic mesh adaptation**

[Loseille et al., AIAA 2007], [Alauzet, IJNMF 2008], [Loseille and Alauzet, IMR 2009]

- **Optimal** control of the interpolation error in  $L^P$  norm :  $\|W - \Pi_h W\|_{L^P(\Omega_h)}$
- **Highly anisotropic** meshes
- Capture **all scales** of the flow
- **Global 2<sup>nd</sup>** of mesh convergence for the mesh adaptation process
- **Early capturing property**: asymptotic convergence is reached faster



## Functional approximation error based mesh adaptation

Deriving the **best mesh** to observe a given functional:

$$J(W) = (g, W) \quad \text{with} \quad \nabla \cdot \mathcal{F}(W) = 0$$

- Depends explicitly on the PDE and numerical scheme
- The goal is to derive the best solution of a PDE

[Venditti and Darmofal, JCP 2003], [Jones et al., AIAA 2006], [Power et al., CMA 2006], [Wintzer et al., AIAA 2008], [Leicht and Hartmann, JCP 2010], ...

## Issues with goal-oriented anisotropic mesh adaptation

- **Anisotropy** is hard to prescribe
- Error estimate are **difficult to obtain** and are **not generic**:
  - Explicit use of the PDE
  - Strong dependency on the numerical scheme
- There is an **over cost** as it requires to compute the adjoint state

# Goal-Oriented Anisotropic Mesh Adaptation

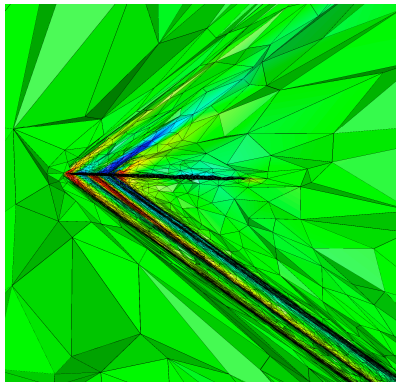
We propose a **goal-oriented anisotropic mesh adaptation**

[Loseille et al., AIAA 2010], [Loseille et al., JCP 2010]

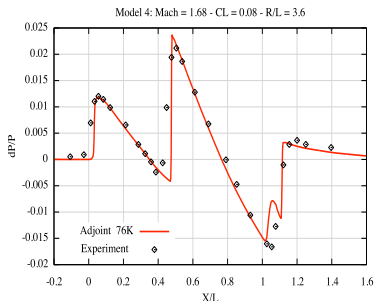
- **Optimal** control of the approximation error in  $L^1$  norm

$$\|J(W) - J(W_h)\|_{L^1(\Omega_h)} \approx \|\nabla W^* \cdot (\mathcal{F}(W) - \Pi_h \mathcal{F}(W))\|_{L^1(\Omega_h)}$$

- **Highly anisotropic** meshes and capture **all scales** of the flow
- **Global 2<sup>nd</sup>** of mesh convergence for the output functional



- **76K** vertices
- **Mean ratio 47**
- **CPU time 23m02s**
- **Mean quotient 1410**





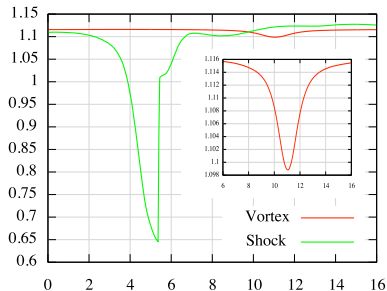
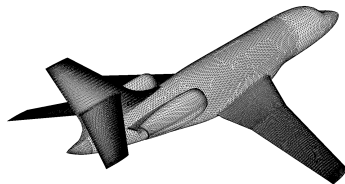
# Wing Tip Vortices Accurate Computation

## Computation of wing tip vortices of a Dassault Falcon Jet:

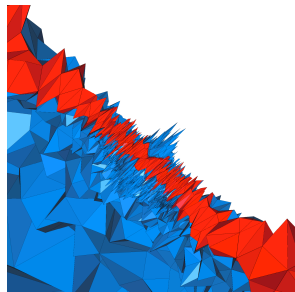
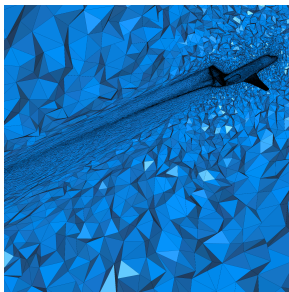
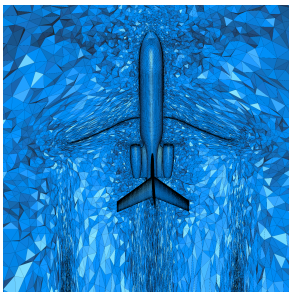
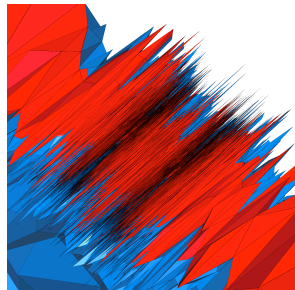
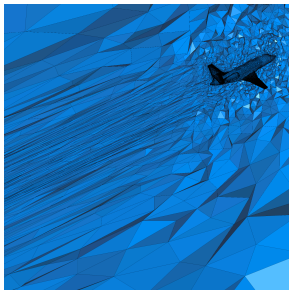
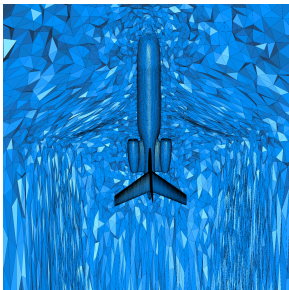
- Adjoint functional :

$$J(W) = \int_{\gamma} \|\nabla \wedge (\mathbf{u} - \mathbf{u}_{\infty})\|_2^2 d\gamma$$

- Adaptation variable : Mach number

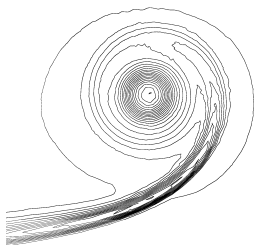


# Comparison between adjoint (top) and feature (bottom) based adapted meshes

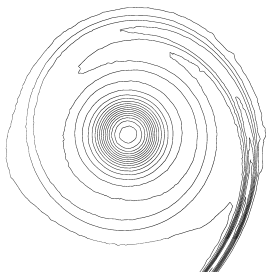


# Comparison between adjoint (top) and feature (bottom) based wing tip vortices

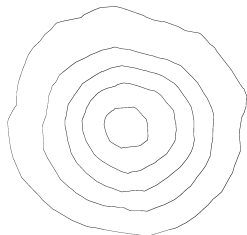
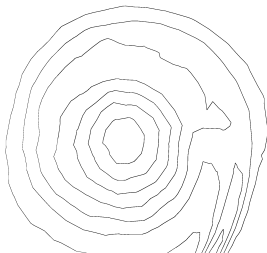
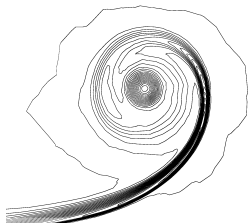
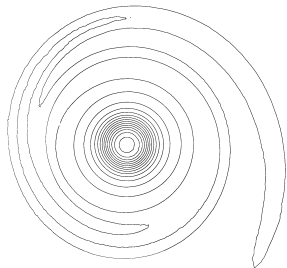
100 m



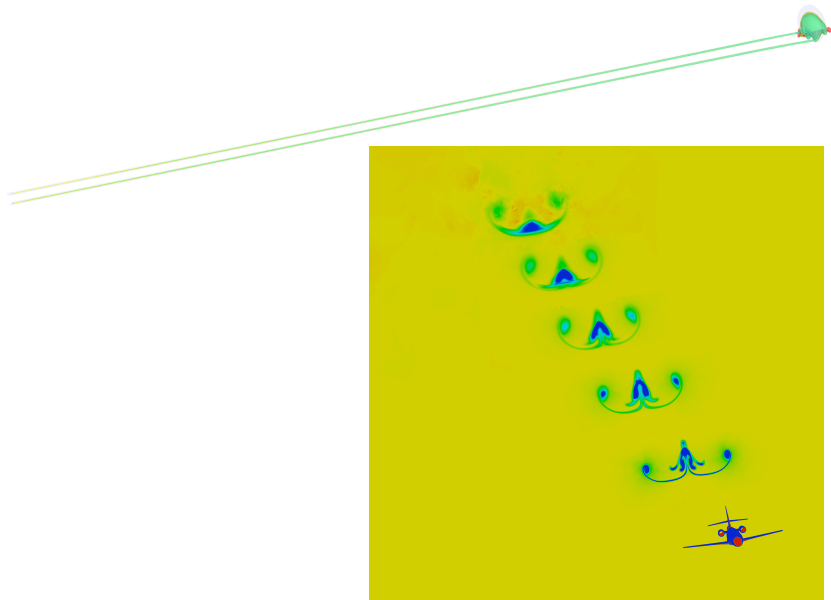
200 m



400 m



Vorticity iso-surfaces :



Let time-dependent function

$$J = \frac{1}{2} \int_{S \times (0, T)} |B \cdot W - b|^2 \quad \text{with} \quad \partial_t W + \nabla \cdot F(W) = 0$$

The extended calculus of variation on  $J$  gives [Alauzet and Pironneau, IJNMF 2012]

$$\delta J = \int_{S \times (0, T)} (B \cdot \overline{W} - b) B \cdot \delta W \quad \text{with} \quad \overline{W} = \frac{1}{2}(W^+ + W^-)$$

Turning to  $\delta W$  in state equation:

$$\partial_t \delta W + \nabla \cdot (\overline{F'(W)} \delta W) = 0 \quad \text{and} \quad \delta W(0) = 0$$

The adjoint equation:

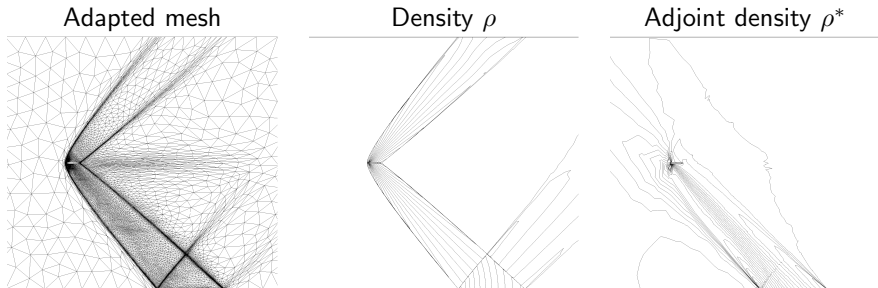
$$\partial_t W^* + \overline{F'(W)} \nabla W^* = 0, \quad W^*(T) = 0 \quad \Rightarrow \quad \int_{\partial \Omega \times (0, T)} W^* \cdot (n \cdot (\overline{F'(W)} \delta W)) = 0$$

Thus, this gives a method to evaluate continuous  $W^*$  and  $\delta J$

$$W^* \cdot (n \cdot (\overline{F'(W)}))|_{\Omega} = (B \overline{W} - b) B^T \quad \Rightarrow \quad \delta J = - \int_{\partial \Omega \setminus S \times (0, T)} W^* \cdot (n \cdot (\overline{F'(W)} \delta W))$$

# Sensitivity of Functionals of Euler Equations with Shocks

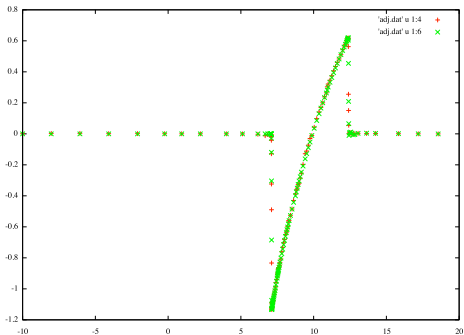
Validation on a NACA0012 airfoil that for **non-smooth flow** discrete adjoint converge to continuous one



The theory on the continuous systems tells that the adjoint is continuous across the shocks but maybe discontinuous elsewhere, including where  $W$  has slip-discontinuities

# Sensitivity of Functionals of Euler Equations with Shocks

Validation on a NACA0012 airfoil that for **non-smooth flow** discrete adjoint converge to continuous one



Comparison between  $(\rho v)^*$  and  $(\rho_h v_h)^*$  on the observation surface  $S$

Conclusion: discrete and continuous calculus of variation agree

## Time-accurate anisotropic mesh adaptation

Deriving the **best space-time mesh** to observe a given functional

$$J(W(\mathbf{x}, t)) \quad \text{with} \quad \partial_t W(\mathbf{x}, t) + \nabla \cdot \mathcal{F}(W(\mathbf{x}, t)) = 0$$

[Lohner and Baum, IJNMF 1992], [De Sampaio et al., CMAME 1993], [Speares and Berzins, IJNMF 1997], [Pain et al., CMAME 2001], [Remacle et al., IJNME 2005], ...

## Issues with time-accurate anisotropic mesh adaptation

- Solution **evolves** in time
  - **Temporal** error occurs
  - The mesh is always **lagging behind** the solution
- **Steady case is applied frequently** in time: each  $n^{\text{th}}$  iterations
  - A lot of solution interpolation  $\implies$  **spoil solution accuracy**
  - A lot of remeshing  $\implies$  **CPU cost increase**
  - **Require a flow solver with dynamic data**
- **4D space-time** anisotropic mesh adaptation



## Time-accurate anisotropic mesh adaptation

Deriving the **best space-time mesh** to observe a given functional

$$J(W(\mathbf{x}, t)) \quad \text{with} \quad \partial_t W(\mathbf{x}, t) + \nabla \cdot \mathcal{F}(W(\mathbf{x}, t)) = 0$$

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We propose a **fixed-point anisotropic mesh adaptation** strategy based on

[Alauzet et al., IMR 2011], [Belme et al., JCP 2012]

- **Optimal** control of the space-time approximation error on  $J$  in  $L^1$  norm

$$\|J(W) - J(W_h)\|_{L^1([0, T] \times \Omega_h)} \approx \|W_t^* (W - \Pi_h W) + \nabla W^* \cdot (\mathcal{F}(W) - \Pi_h \mathcal{F}(W))\|_{L^1([0, T] \times \Omega_h)}$$

- **Global fixed-point** mesh adaptation algorithm
- **Conservative** solution transfer

Solve this problem in the continuous mesh framework

From [Loseille and Alauzet, SINUM 2010], after discarding BT:

$$\begin{aligned}\mathbf{E}(\mathbf{M}) &= \int_0^T \int_{\Omega} |W_t^*| |W - \pi_{\mathcal{M}} W| \, d\Omega \, dt + \int_0^T \int_{\Omega} |\nabla W^*| \cdot |\mathcal{F}(W) - \pi_{\mathcal{M}} \mathcal{F}(W)| \, d\Omega \, dt \\ &= \int_0^T \int_{\Omega} \text{trace} \left( \mathcal{M}^{-\frac{1}{2}} \mathbf{H} \mathcal{M}^{-\frac{1}{2}} \right) \, d\Omega \, dt\end{aligned}$$

$$\text{with } \mathbf{H} = \sum_{i=1}^5 \left( |W_{i,t}^*| |H(W_i)| + \sum_{j=1}^3 |\nabla_{x_j} W_i^*| |H(\mathcal{F}_{x_j}(W_i))| \right)$$

## Mesh Optimization Problem

Find  $\mathbf{M}_{go} = (\mathcal{M}_{go}(\mathbf{x}))_{\mathbf{x} \in \Omega}$  of complexity  $N_{st}$  such that

$$\mathbf{E}_{go}(\mathbf{M}_{go}) = \min_{\mathbf{M}} \int_0^T \int_{\Omega} \text{trace} \left( \mathcal{M}^{-\frac{1}{2}} \mathbf{H} \mathcal{M}^{-\frac{1}{2}} \right) \, d\Omega \, dt$$

$$\text{under constraint } \mathcal{C}_{st}(\mathbf{M}) = N_{st} = \int_0^T \tau(t)^{-1} \left( \int_{\Omega} d_{\mathcal{M}}(\mathbf{x}, t) \, d\mathbf{x} \right) \, dt$$

# G-O EE for Unsteady Compressible Euler Equations

Solve this problem in the continuous mesh framework

Two steps resolution:

1. Spatial minimization for fixed  $t$
2. Temporal minimization

Spatial minimization for fixed  $t$

$$\mathcal{M}_{go}(\mathbf{x}, t) = N(t)^{\frac{2}{3}} \mathcal{K}(t)^{-\frac{2}{3}} (\det \mathbf{H}(\mathbf{x}, t))^{-\frac{1}{5}} \mathbf{H}(\mathbf{x}, t)$$

Temporal minimization for specified  $\tau$  [Belme et al., JCP 2012]

$$\mathcal{M}_{go}(\mathbf{x}, t) = N_{st}^{\frac{2}{3}} \left( \int_0^T \tau(t)^{-\frac{2}{5}} \mathcal{K}(t) dt \right)^{-\frac{2}{3}} \tau(t)^{\frac{2}{5}} (\det \mathbf{H}(\mathbf{x}, t))^{-\frac{1}{5}} \mathbf{H}(\mathbf{x}, t)$$

with  $\mathcal{K}(t) = \left( \int_{\Omega} (\det \mathbf{H}(\bar{\mathbf{x}}, t))^{\frac{1}{5}} d\bar{\mathbf{x}} \right)$

**Remark:** A temporal minimization for time sub-intervals has also been achieved

# Discrete Unsteady Adjoint Resolution

Forward in time state equation:

$$W_h^n = W_h^{n-1} + \delta t^n \Phi_h(W_h^{n-1})$$

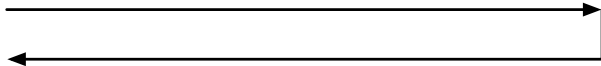
Consider a time-dependent functional:

$$J(W) = \int_0^T \int_{\Gamma} j_{\Gamma}(W(\mathbf{x}, t)) d\mathbf{x} dt$$

Backward in time adjoint equation:

$$W_h^{*,n-1} = W_h^{*,n} + \delta t^n \frac{\partial J_h^{n-1}}{\partial W_h^{n-1}}(W_h^{n-1}) - \delta t^n (W_h^{*,n})^T \frac{\partial \Phi}{\partial W_h^{n-1}}(W_h^{n-1})$$

Solve state forward:  $\Psi(W) = 0$



Solve adjoint state backward:  $\Psi^*(W, W^*) = 0$

# Discrete Unsteady Adjoint Resolution

## Problematics

- Computing  $W^{*,n-1}$  at time  $t^{n-1}$  requires the knowledge of state  $W^{n-1}$  and adjoint state  $W^{*,n}$

⇒ The knowledge of **all states**  $\{W^n\}_{n=1,N}$  is needed

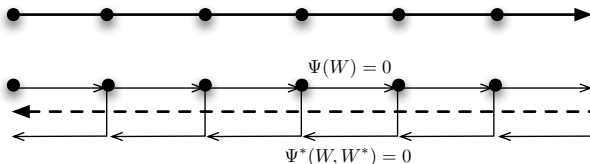
⇒ Large memory storage effort in 3D

( $10^6$  vertices &  $10^3$  iterations request 37.25 Gb)

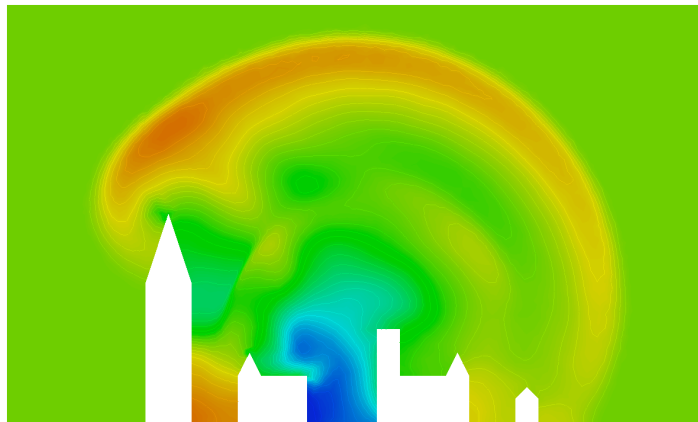
Reduce the memory storage effort by:

- **out-of-core storage of checkpoints**  
⇒ **recomputing effort of the state  $W$**
- **state interpolation between two memory storage**  
⇒ **slight loss of accuracy**

Solve state once to get checkpoints  $\Psi(W) = 0$



# A Movie to Illustrate the Approach



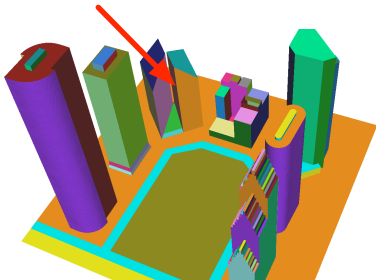
# High-Fidelity Blast Computation

## Application: High-Resolution Blast Wave Prediction

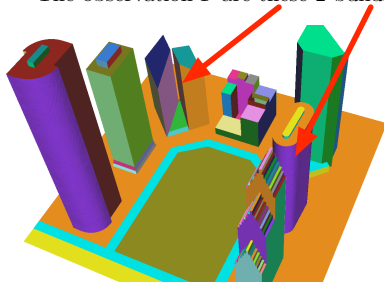
Goal-oriented time-accurate aniso. mesh adaptation on pressure impulse

- From 40K to 1million vertices
- From 200K to 6 million tets
- Mean aniso ratio  $\approx 100$
- Mean aniso quotient  $\approx 3000$

The observation  $\Gamma$  is this building



The observation  $\Gamma$  are these 2 buildings

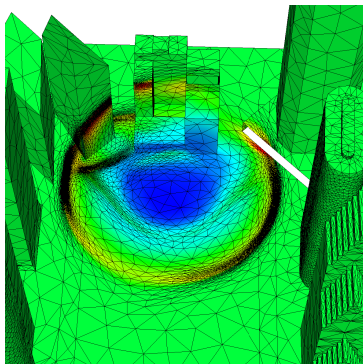
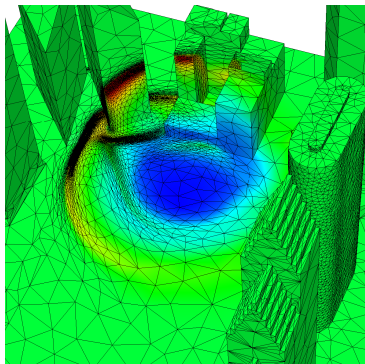


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Goal-oriented time-accurate aniso. mesh adaptation on pressure impulse

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## Last Decades:

- 1990's Automated 3D unstructured mesh generation  
⇒ **inviscid** computation on **complex** geometries
- 2000's Automated 3D BL mesh generation  
⇒ **viscous** computation on **complex** geometries
- 2010's Automated highly anisotropic mesh adaptation  
⇒ High-fidelity **inviscid** computation on **complex** geometries

## Current and Next Decades:

- Highly anisotropic mesh adaptation for **viscous** flows  
⇒ Coupling structured and unstructured mesh adaptation
- Highly anisotropic mesh adaptation for **very-high order** solvers  
⇒ Curved meshes, very-high order error estimates
- Highly anisotropic mesh adaptation for **moving geometries** problem  
⇒ Adaptive moving mesh methods, dynamic error estimates

Thank you for your attention

